

SENSORS AND CERTIFIABLE HYBRID ARCHITECTURES
FOR SAFER AVIATION IN ICING ENVIRONMENT

# Collins IWT Capabilities and Icing Characterization

SAE AC-9C Meeting

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Virtual – 22 April 2021

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This document does not contain any export controlled technical data



# **Outline**

- SENS4ICE project overview
- **♦** SENS4ICE Icing Wind Tunnel Requirements
- Overview of Collins test facility capabilities
- Collins facility characterization and calibration
  - App. C calibration instrumentation and methodology
  - App. O characterization instrumentation and methodology with pictures
  - ♠ SENSAICE CCP and Nevzorov measurements at Collins
- Collins Ice Differentiator overview
- Summary and Future Work





# **SENS4ICE Consortium Partners**

















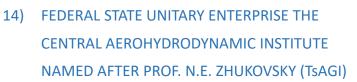


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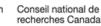
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## **SENS4ICE Overview**

**Objective:** Increase flight safety in icing conditions and especially for the SLD conditions

### **Problem**

Detect icing conditions (including App. O/ SLD icing) – detection very challenging

### **Solution**

Hybrid approach – fusion of input data: sensor(s) and indirect detection

### **Benefits**

- Operational benefits: activate anti-/de-icing, avoid/ leave icing conditions
- ♠ Certification process benefits flights in App. O/ SLD icing

Project duration: JAN 2019 - DEC 2022 (project extension

expected)

**Coordinator**: DLR





# **SENS4ICE Icing Wind Tunnel Requirements**

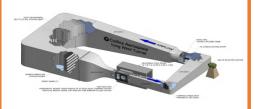
- One of the main objectives of SENS4ICE is to develop and mature different ice detection technologies with a focus on App O conditions
- There are 10 different ice detectors being developed by the project partners
- Most of these ice detectors target detection and discrimination between App. O and App. C conditions
- Detectors performance need to be tested in relevant environment at icing wind tunnel
- Extensive icing wind tunnel tests are planned in the project
  - 28 weeks of testing distributed over four facilities
  - Parallel testing planned to meet project timeline
- Testing will provide data for technology selection towards flight test





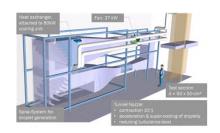
# **Overview of SENS4ICE IWT Capabilities**

### Collins Aerospace, USA



- 5-147 micron droplets
- LWC between 0.1 and 3 g/m3
- Temperature 0°C to -30°C
- Sustained speed 13-103 m/s
- Test section: 152x56x112 cm3
- Calibrated per SAE ARP 5905
- Compliant with AS9100C
- Controls and power supplies can simulate aircraft controls

### TU Braunschweig, Germany



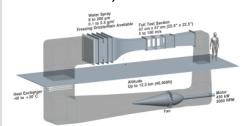
- MVD 9-60 micron droplets
- LWC between 0.1 and 1.5 g/m3
- Temperature 30°C to -20°C
- Sustained speed 10-40 m/s
- Test section: 150×50×50 cm3
- Calibrated per SAE ARP 5905
- Short spray transients ~ 15s
- Bi-modal SLD and mixed phase capability

#### TsAGI, Russia



- 10-90 micron droplets
- LWC between 0.5 and 6 g/m3
- Temperature down to -40°C
- Sustained speed up to 150 m/s
- Test section: 300×100×100 cm3
- PDI Artium 2D PSD calibration
- LWC calibration with EIV-2K
- High speed camera with longfocus microscope

#### NRC, Canada



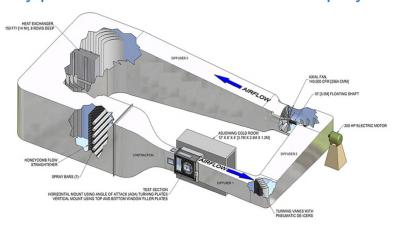
- 8-200 micron droplets
- LWC between 0.1 and 2.5 g/m3
- Supercooled Water: 10 to > 200 µm (incl. SLD bi-modal)
- Temperature +30°C to -40°C
- Sustained speed 5-100 m/s
- Test section: 57×57 cm2 (52x33 cm2 with insert)
- Sea level < Altitude < 40,000ft
- Calibrated per SAE ARP 5905





# **Collins Icing Wind Tunnel Overview**

- The Collins Aerospace Icing Wind Tunnel (IWT), constructed in 1988, is a closed-loop refrigerated tunnel measuring 40' x 70' (12.2 m x 21.3 m) overall. Tunnel had series of upgrades including control system, structure and heat exchange (additional tonnage).
- The IWT has been used for the development and design validation testing of a variety of aerospace models including airfoils, engine nacelle sectors, struts, and inlet guide vanes.
- The IWT utilizes an external 200 hp electric motor driving a 79" (2.0 m) diameter axial fan to provide wind velocity, a 70 ton capacity refrigeration system for cooling, and a series of deionization tanks to continuously produce DI water for use in the spray bar system.

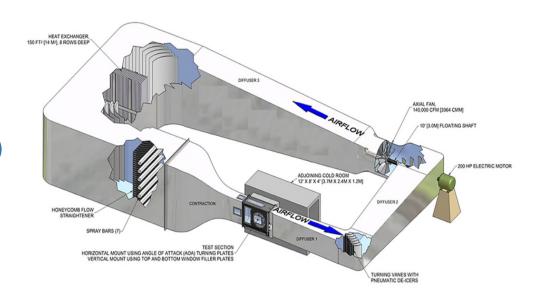






# **Collins Icing Wind Tunnel Capabilities**

- **Test Section Dimensions:** 
  - 22" (0.6 m) wide x 44" (1.1 m) high x 60" (1.5 m) long
- Temperature Range:
  - -22°F to 32°F (-30°C to 0°C)
- Test Section Airspeed Range:
  - 30 mph to 230 mph (26 kn to 200 kn)
- App C LWC Range:
  - $0.15 \text{ g/m}^3 \text{ to over } 3.00 \text{ g/m}^3$
- App C MVD Range:
  - 15 μm to over 40 μm





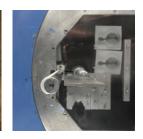


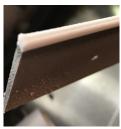
# **Collins IWT - Appendix C Calibration**

- App C calibration procedure based on the best practices outlined in SAE ARP5905.
- LWC calibration is performed using an icing blade.
- MVD calibration is performed using a PDI laser manufactured by Artium Technologies.
  - When configured with a 500mm focal length lens, the PDI system has an effective measurement range of 3µm -300µm

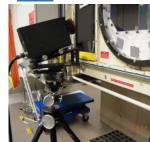
### Icing Blade:

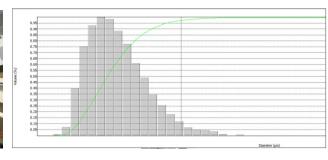






### PDI:





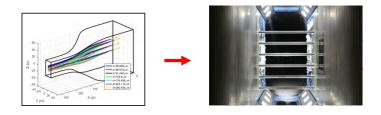




# **Collins IWT - Appendix O Characterization**

- The IWT's Appendix O capabilities were characterized in preparation for the SENS4ICE testing campaign.
  - Process for developing App. O was an iterative loop, computer simulation was used to inform which conditions to actually test in the IWT and vice versa.
- LWC characterization was performed using an SEA Multi-Flement Probe.
- MVD characterization was again performed using the PDI laser manufactured by Artium Technologies.
  - For App. O conditions, a lens with a 1000mm focal length was used.
  - This augmented the PDI's measurement capabilities to an effective range of 6µm - 570µm.

### **App O Characterization Process:**



### **SEA Multi-Element Probe:**





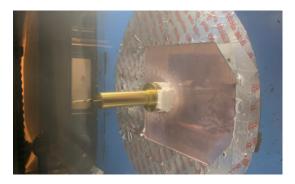




# **SENS4ICE Testing – CCP & Nevzorov Probes**

- The CCP and Nevzorov probes are serving as reference probes for the icing wind tunnels participating in the consortium
- The Nevzorov probe was tested at Collins for 2 days.
  - The entirety of the App C and App O test matrices were completed.
- The CCP probe was tested at Collins over the course of 3.5 days.
  - The entirety of the App. C and App O. test matrices were completed.
  - During the test only one of the CCP probe's measurement volumes could be positioned in the centerline of the tunnel at a time.
  - For that reason, many of the test conditions were repeated in two configurations so that data could be collected with each measurement volume centered in the tunnel.

### **Nevzorov Probe:**



**CCP Probe:** 





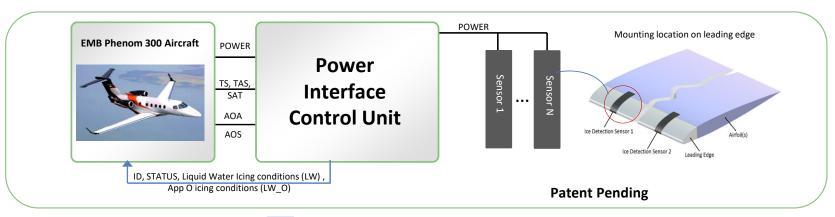


# **Collins Ice Differentiator Technology**

### **Sensor Detection Concept**

SENS4ICE, EU-funded project, Grant Agreement No 824253

- Ice detection based on thermal response to a heat impulse that changes from dry to icing conditions
- Heat flux variation measured using composite heater
- System is made of two components: a sensing element (heater) and a power interface control unit
- Power interface unit analyses measurements and makes recommendations on icing conditions Dry/App. C/App. O

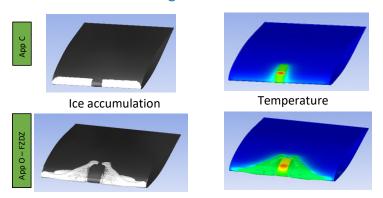




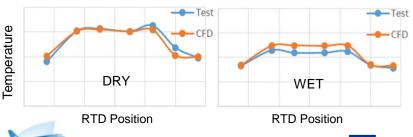
# **Collins Ice Differentiator Achieved TRL 4**

### **NUMERICAL ANALYSIS**

- Numerical analysis used to define sensor design parameters
- Model validated against IWT test measurements



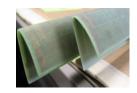
### Model validation against IWT test data



SENS4ICE, EU-funded project, Grant Agreement No 824253

#### LAB AND IWT TESTING

- Sensor prototype power delivery on controller request and data acquisition tested
- Integration tests performed using a thermal chamber

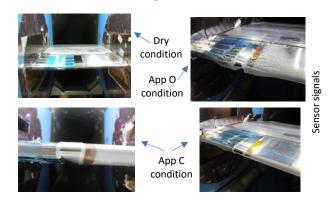








Four rounds of IWT tests completed at Collins IWT – OH and NRC IWT
 View of sensor during IWT tests



22/04/2021

# **Summary and Future Work**

- Collins IWT facility offers App C and App O capabilities
- ♦ Four weeks of SENS4ICE testing were completed at Collins IWT
- ♦ CCP and Nevzorov successfully tested in the Collins IWT
- ♦ Collins IDS (Ice Differentiator Sensor CIDS) tested during 120 hours at Collins IWT
- Collins IDS tested at NRC icing tunnel for additional test points
- ♦ Collins IDS achieved TRL4 following IWT tests and is capable of detection and differentiation between Dry, App. C and App O. icing conditions
- Icing wind tunnel tests on vertical fin planned in Q4 to prove scalability of the technology
- Preparations for potential flight test with Embraer



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