



# SENS4ICE

SENSORS AND CERTIFIABLE HYBRID ARCHITECTURES  
FOR SAFER AVIATION IN ICING ENVIRONMENT

## Short Range Particulate (SRP)

FINAL DISSEMINATION EVENT OF SENS4ICE PROJECT

Pavel Badin - Honeywell

Directorate General for Research and Innovation, Brussels, Belgium – 29 November 2023

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## Sensor Description



# SRP – Short Range Particulate

Honeywell Aerospace

- 💧 Detection type: Atmospheric sensor
- 💧 Physical principle: Collecting backscattered light from particles
- 💧 Sensor high-level output description
  - 💧 App C flag, App O flag
  - 💧 LWC [ $\text{g}/\text{m}^3$ ], MVD [ $\mu\text{m}$ ], DV90 [ $\mu\text{m}$ ], DV99 [ $\mu\text{m}$ ], DMAX [ $\mu\text{m}$ ]
- 💧 Air sensor specifications
  - 💧 Size: 290 x 180 x 130 x mm (\*\*\*)
  - 💧 Weight: 4.7 kg (\*\*\*)
  - 💧 Power: 300 W
- 💧 Equipment TRL 3 at project start, TLR 6 now
  - 💧 App C detection (TRL6), App O detection (TRL5), App O / C discrimination (TRL5), Ice Crystals Detection (TRL4)
- 💧 Icing Wind Tunnel tests: 2016, 2017, 2021
- 💧 Flight tests: 2016, 2020, 2023



*1<sup>st</sup> gen. optical sensor  
direct ptcl. sensing: 2 - 42  $\mu\text{m}$*



*2<sup>nd</sup> gen. optical sensor  
(developed under SENS4ICE)  
direct ptcl. sensing: 50 - 1000  $\mu\text{m}$   
background signal sensing: 5 - 50  $\mu\text{m}$*

*(\*\*\*) applicable to technology demonstrator only, sensor size and weight to be reduced*

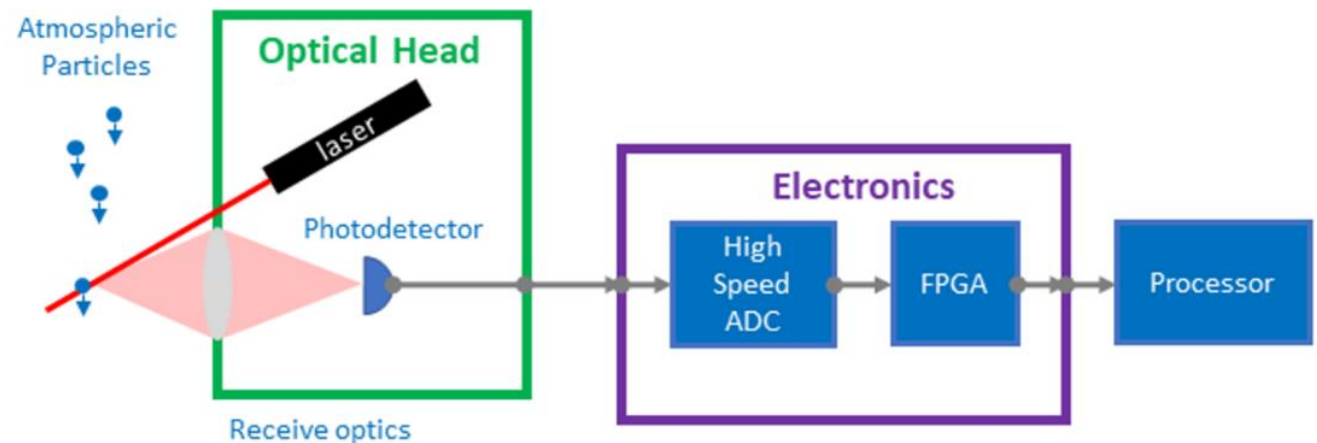
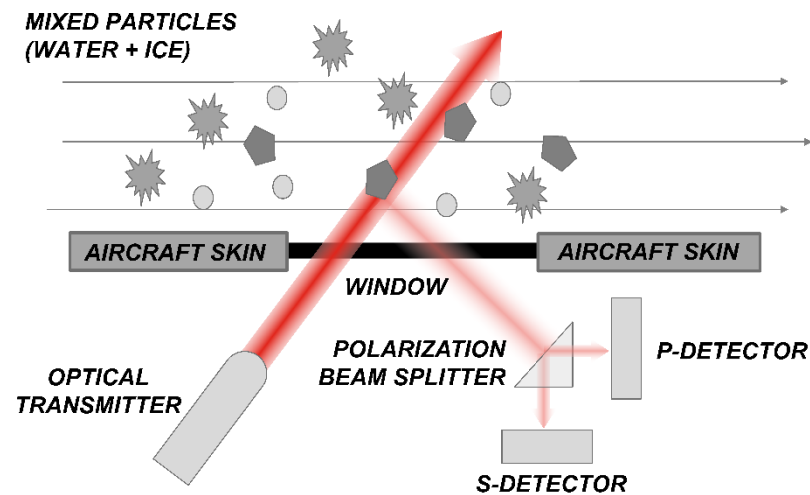


# SRP: Technology Overview

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## Optical sensor detecting particles through polarized backscatter

- Extract data two ways
  - Direct particle by particle measurements (scattering from a single particle)
  - Aggregate particle scatter through shifts in the background signal

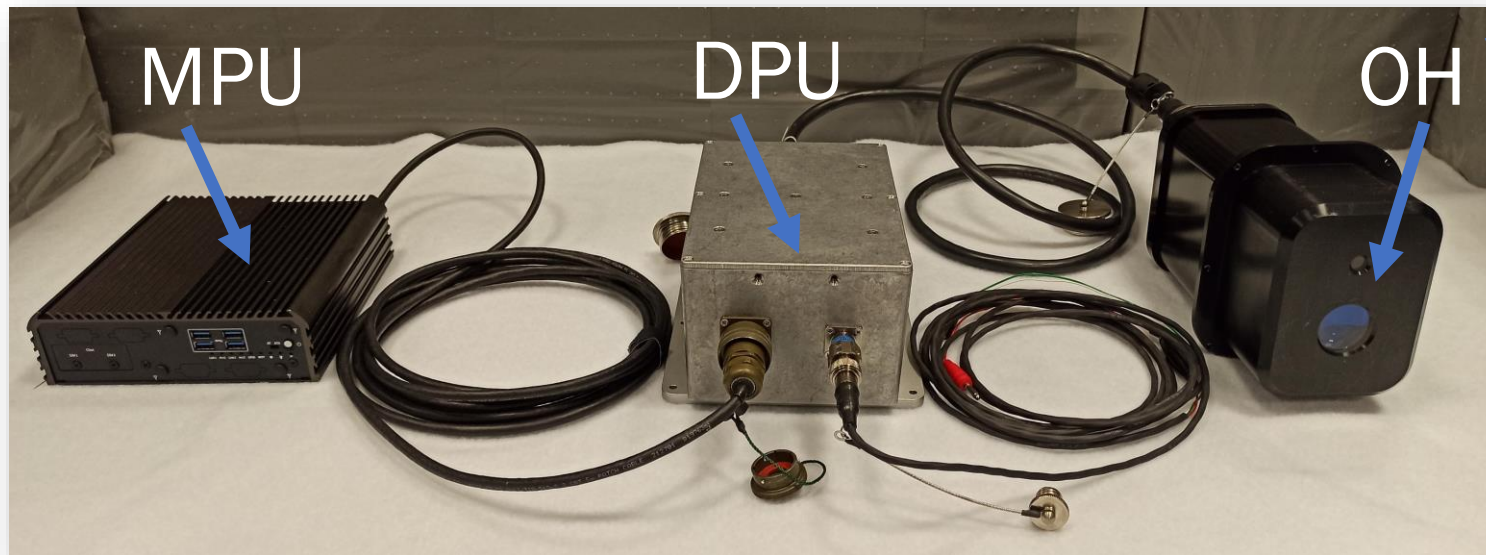


# SRP: Technology Overview

Honeywell Aerospace

## Sensor prototype description

- MPU: Main Processing Unit
- DPU: Digitalization and Pre-processing Unit
- OH: Optical Head



*SRP installed on Embraer Phenom 300  
© 2023 Embraer*





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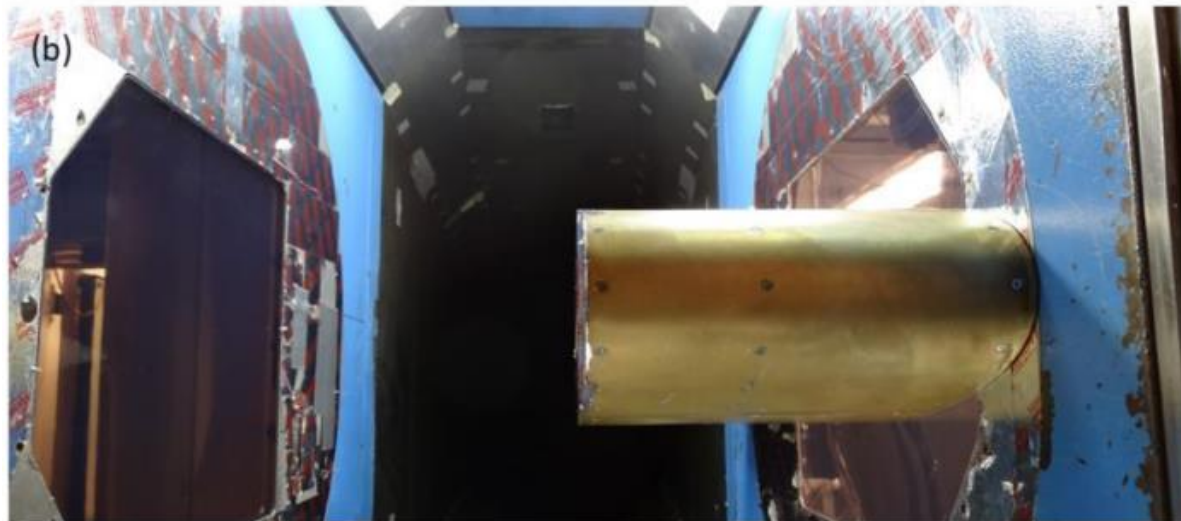
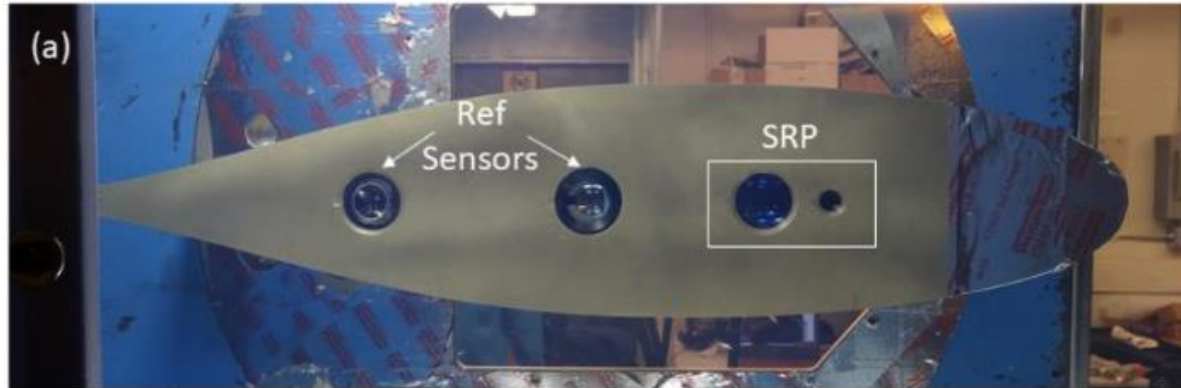
## Icing Wind Tunnel tests



# SRP: Icing Wind Tunnel tests

Honeywell Aerospace

IWT Collins Facility (2021), App. C&O conditions



# SRP: Icing Wind Tunnel tests

## Honeywell Aerospace

- Excellent performance on response time metric across both icing appendices
- Parameter measurement accuracy differed between the two appendices
  - Good match with tunnel values for App. C
  - Higher Error for App. O – discussed in subsequent slide

Test	Test Points Detected within Response Time [%]	Average MVD Error [%]	Average LWC Error [%]
<b>Appendix C Test Points</b>	100%	14%	28%
<b>Appendix C Repeat Points</b>	100%	15%	27%
<b>Appendix O Test Points</b>	100%	41%	67%
<b>Appendix O Repeat Points</b>	100%	24%	59%



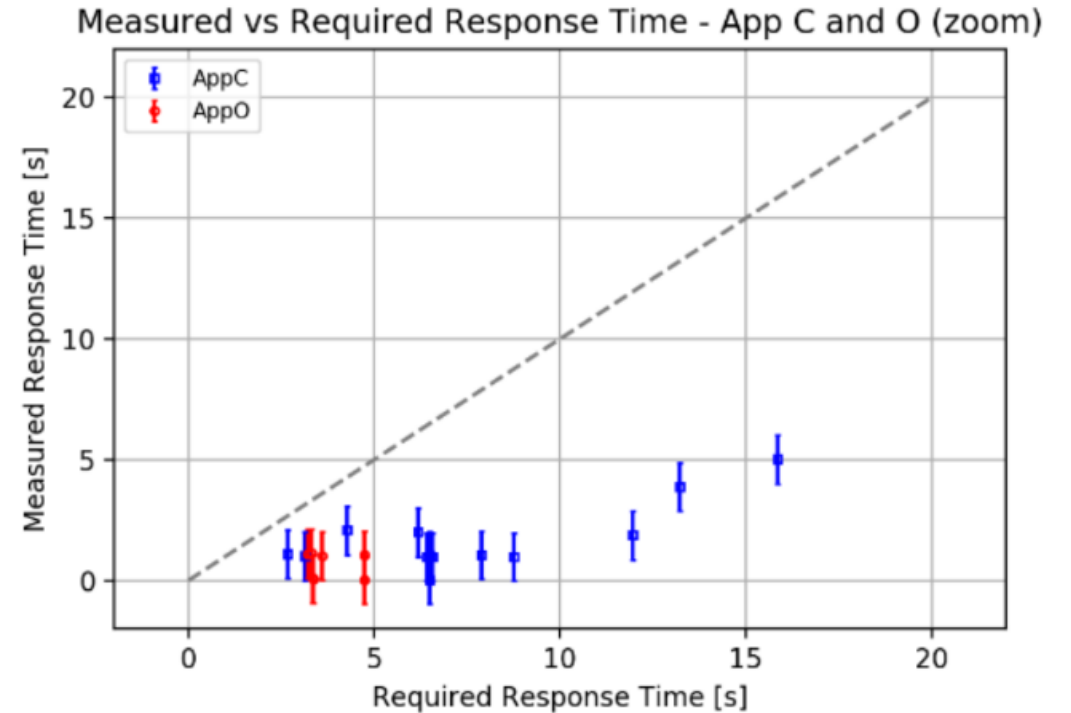
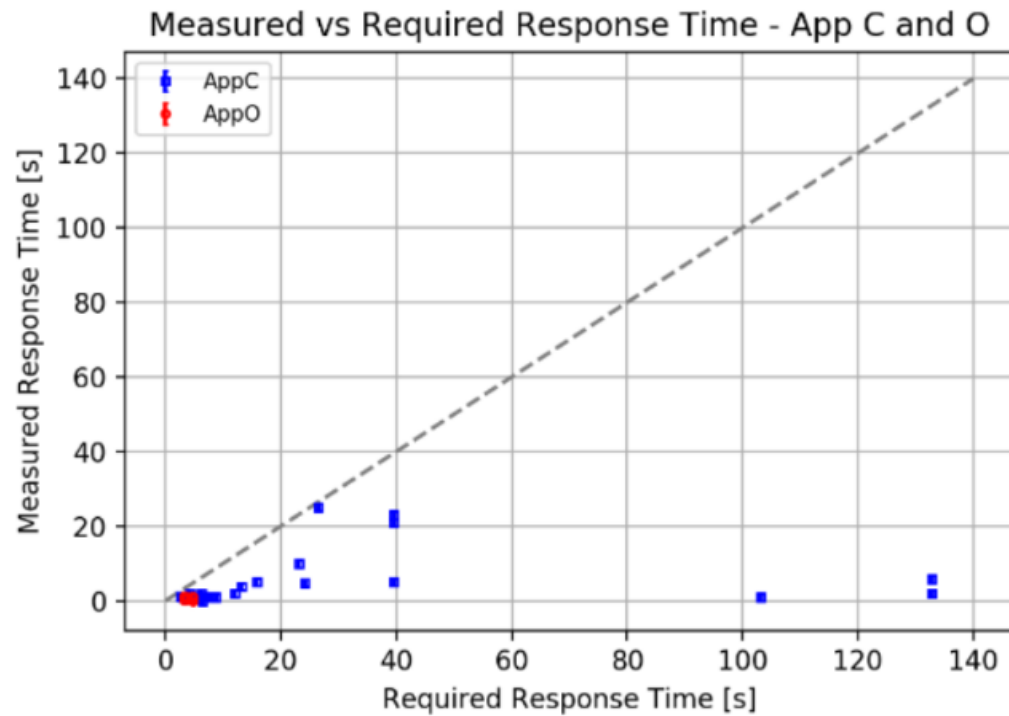


# SRP: Icing Wind Tunnel tests

Honeywell Aerospace

## Response time

- Sensor meets SAE AS5498B (or equivalently ED-103B from EUROCAE) requirements on response time

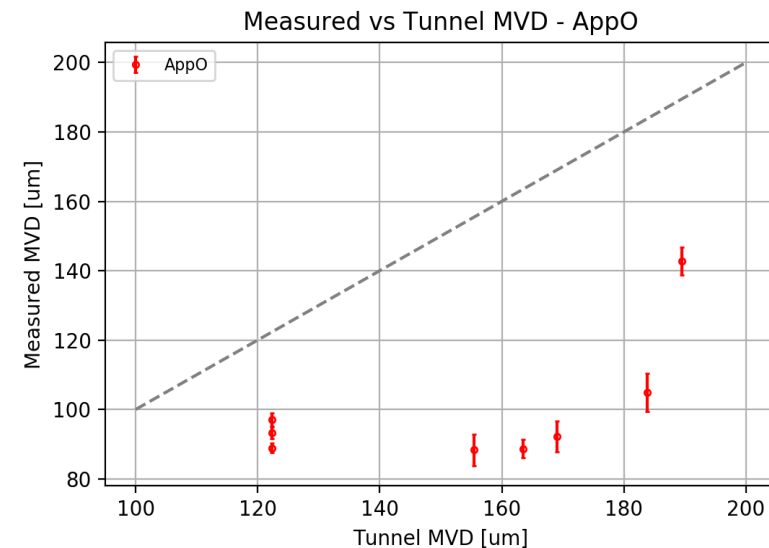
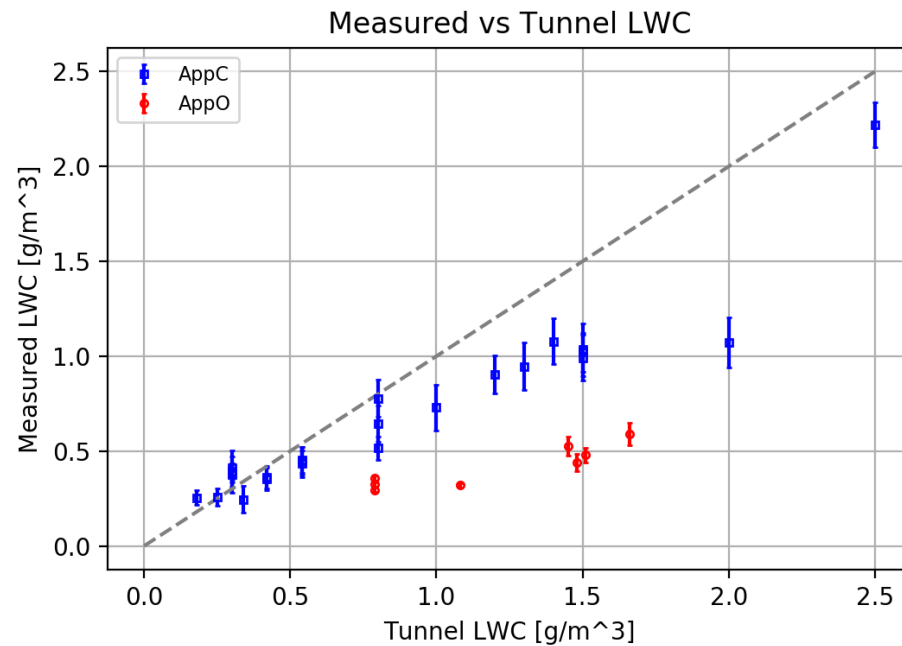
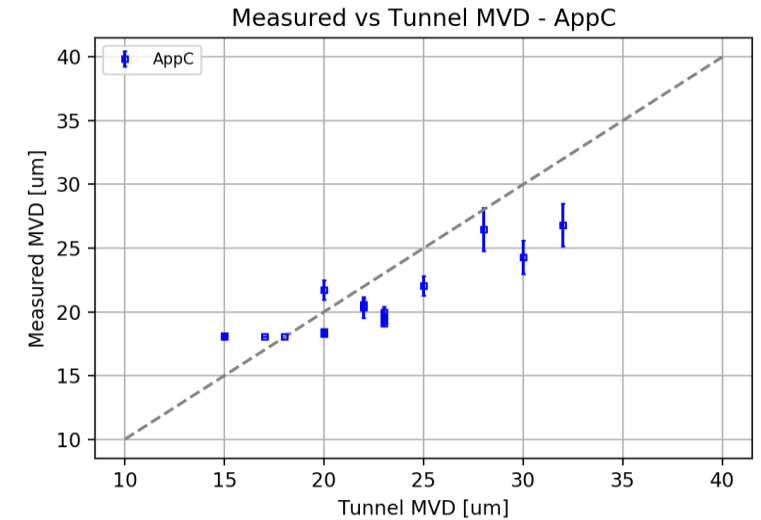


# SRP: Icing Wind Tunnel tests

Honeywell Aerospace

## MVD / LWC Measurements

- Appendix C tracks well for both MVD and LWC
- Larger errors in Appendix O resulting from loss of signal in IWT vs lab calibration
  - Window fogging is expected the root-cause of the issue. Mitigations being developed





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## Flight Test Campaign



# SRP: Flight Test Campaign

Honeywell Aerospace

## Fight test preparation

- Flight test campaign executed late Feb - early Mar in North America
- Prior to flight test, sensor went through flight worthiness
  - Qualified against a subset of DO-160G tests, including temperature, vibration / shock, power input, lightning, and ESD



# SRP: Flight Test Campaign

Honeywell Aerospace

## Flight Test Execution

- Embraer's Phenom 300 optical sensor integration
  - Communication issue between sensor & aircraft was resolved without any impact on the campaign
- **Flight Test Campaign**
  - Eight flights for which reference data were successfully measured
  - Appendix C and Appendix O conditions encountered multiple times
  - Optical sensor data collection successful
- **Sensor data analysis and preliminary results**
  - Performance compared with reference data provided by DLR
  - Good results for events in which particulate MVD > 25 microns
  - Sensor underestimates TWC for events in which particulate MVD < 15-20 microns
  - Detailed results provided for flights 1476 and 1481



*1<sup>st</sup> gen. optical sensor  
direct ptcl. sensing: 2 - 42 um*

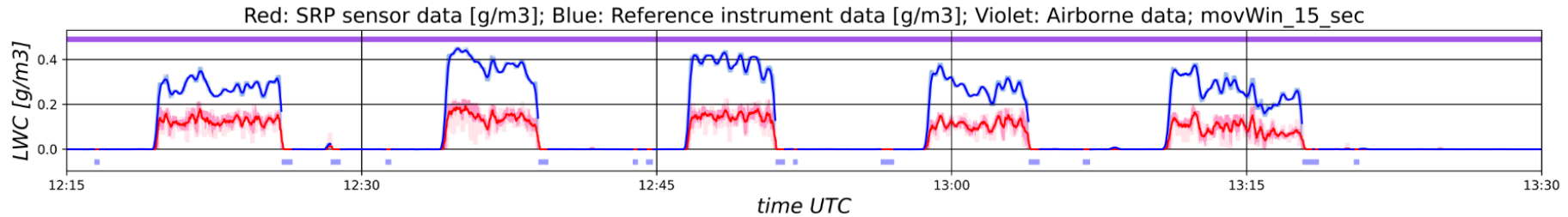


*2<sup>nd</sup> gen. optical sensor  
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background signal sensing: 5 - 50 um*

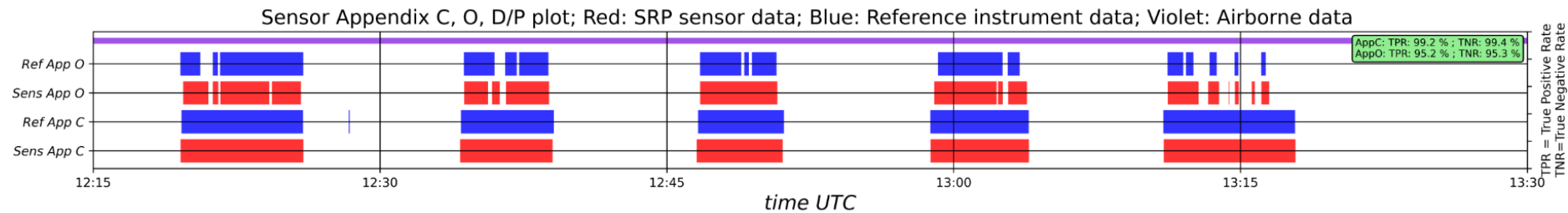


# SRP: Flight Test Campaign (Flight 1476)

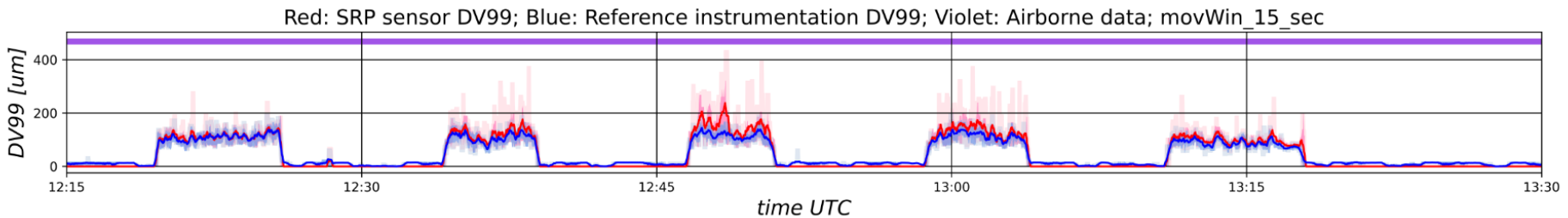
## Honeywell Aerospace



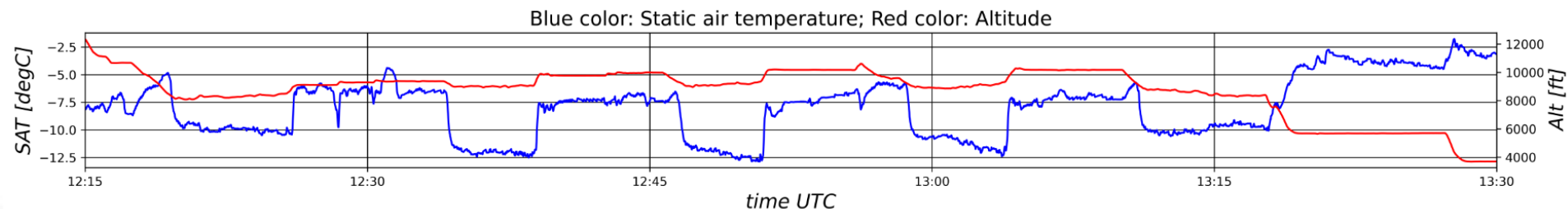
High correlation between reference and optical sensor data



App. C and App. O flags are well matched with reference instrumentation data

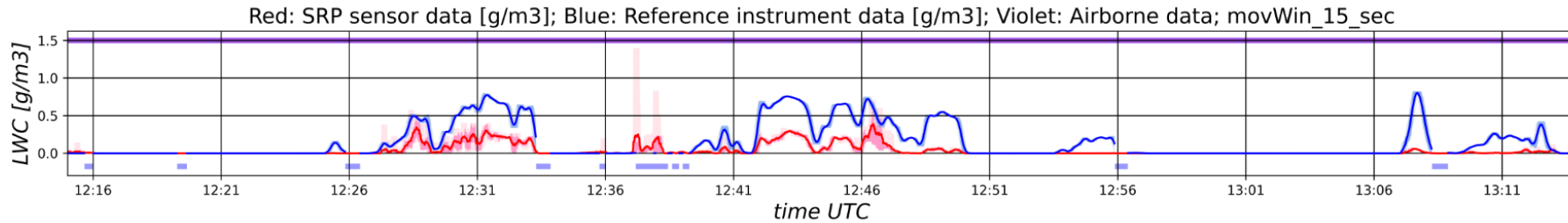


DV99 values are well matched with reference instrumentation data

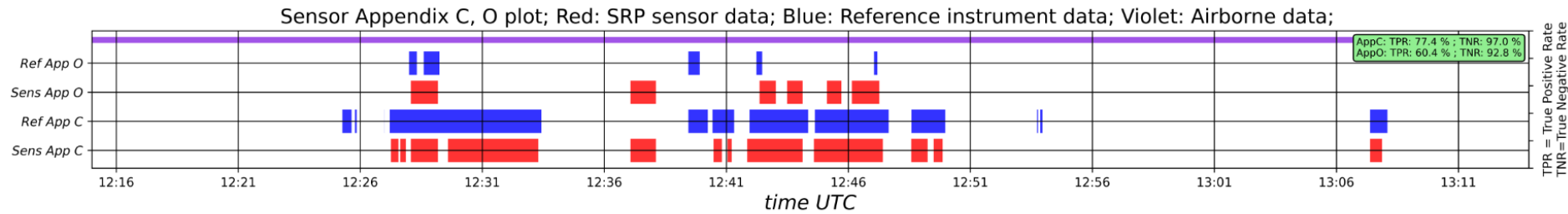


# SRP: Flight Test Campaign (Flight 1481)

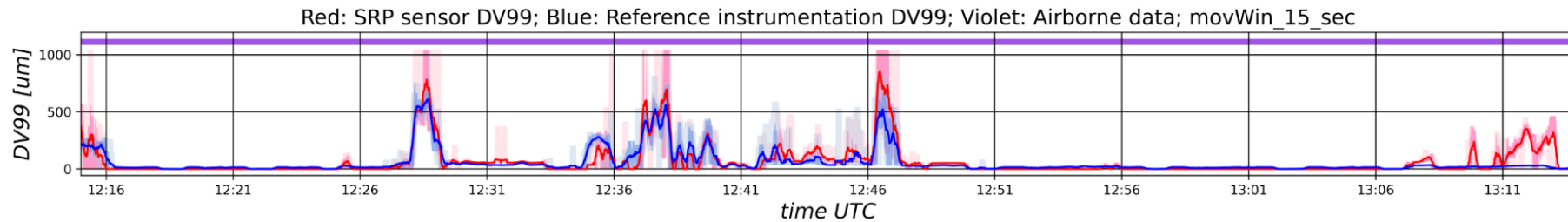
Honeywell Aerospace



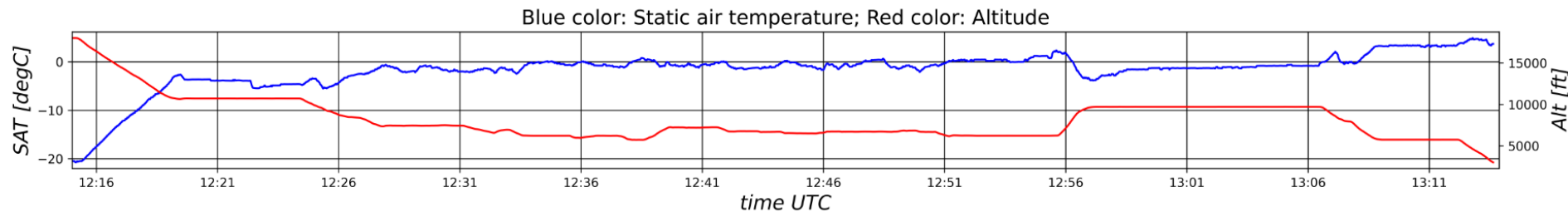
High correlation between reference and optical sensor data



App. C flag is well matched with reference instrumentation data. False alarm presence in App. O flag estimation



DV99 values are well matched with reference instrumentation data



# SRP: Conclusions and Outlook

## Honeywell Aerospace

### Conclusions

- High measurement performance for events in which particulate MVD > 25 microns
- Small particulate detection successfully evaluated under previous program (1<sup>st</sup> gen. optical head)

### Next gen. optical sensor development

- Design single sensor covering icing appendixes C, D, O
- Develop ash / dust / sand sensing functionality

### Use-cases

- Safety, autonomy, situation awareness
- Reduce fuel consumption, reduce CO<sub>2</sub> emissions
- Predictive health maintenance





# SRP: Conclusions and Outlook

## Honeywell Aerospace

### Sensor design update

- Current state
  - 1st generation sensor: Appendix C; 2nd generation sensor: Appendix O, D/P
- Goal
  - Merge both designs into single sensor unit
  - Reduce sensor complexity, size and weight to meet customer requirements

### Appendix D/P characterization

- Evaluate sensor capability to differentiate Liquid Water Content and Ice Water Content
  - Specifically, particles of following diameter: 50 – 1500 microns

### Certification

- Define steps required for sensor certification
- Validate if IWT shattered ice crystals are optically equivalent to natural ice crystals

### Volcanic ash / dust / sand detection

- Develop & Integrate ash / dust / sand sensing capability into the sensor



# SRP: Acknowledgment

## Honeywell Aerospace

- Gratitude to Collins Aerospace and National Research Council Canada for Icing Wind Tunnel test preparation and execution
- Thank you to Embraer, DLR and all SENS4ICE partners for US flight test preparations and execution
- Recognition to DLR and L-Up for project management, data analysis and continuous support

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